



Hyderabad | Delhi | Bhopal | Pune | Bhubaneswar | Bengaluru | Lucknow | Patna | Chennai | Vijayawada | Visakhapatnam | Tirupati | Kukatpally| Kolkata

H.O: 204, II Floor, Rahman Plaza, Opp. Methodist School, Abids, Hyderabad-500001, Ph: 040-23234418, 040-23234419, 040-23234420, 040 - 24750437

# ESE- 2018 (Prelims) - Offline Test Series MECHANICAL ENGINEERING

# Test-1

# **SUBJECT:** FLUID MECHANICS & TURBO MACHINERY - SOLUTIONS

## 01. Ans: (b)

Sol: From the top tube containing alcohol, 11 kPa + (0.9)(10)(3.3)

$$= 40.7 \text{ kPa} = (P_B)_{abs} = (P_A)_{abs}$$

So,

$$(P_A)_{gauge} = (P_A)_{abs} - (P_{atm})$$
  
= 40.7 - 93 = -52.3 kPa

Again,

From the U-tube, we can write

 $(P_B)_{abs} + 0.9 \times 10(y + h) - 13.6 \times 10 \times h = 0$  $(P_B)_{abs} + 9y + 9h - 136h = 0$ 

Or, 127 h = 40.7 + 9×1.6  
= 40.7 + 14.4  
= 55.1  
or, h = 
$$\frac{55.1}{127}$$
 m = 0.434 m = 43.4 cm

02. Ans: (b)

Sol:



With piston alone, let pressure on the piston =  $P_P$ 

The manometric equation gives,

 $P_{\rm P} - \gamma_{\rm oil} h_1 \sin 30^\circ = 0 \qquad (1)$ 

With weight added, pressure P<sub>P</sub> increases to

$$P'_P$$
 where  $P'_P = P_P + \frac{W}{A_P}$ 

where, A<sub>P</sub> is piston area and manometric equation becomes

$$P'_{\rm P} - \gamma_{\rm oil} (h_1 + 0.20) \sin 30^\circ = 0 -----(2)$$

Subtract equation (1) from equation (2)

$$P'_{P} - P_{P} - \gamma_{oil} [h_{1} + 0.20 - h_{1}] \sin 30^{\circ} = 0$$

ACE Engineering Academy



or, 
$$\frac{W}{A_P} = \gamma_{oil} \times 0.2 \times \sin 30^{\circ}$$

W = A<sub>P</sub> × 
$$\gamma_{oil}$$
 × 0.2 ×  $\frac{1}{2}$  =  $\frac{\pi}{4}$  × 0.2<sup>2</sup> × 9.5 × 0.2 ×  $\frac{1}{2}$   
= 9.5  $\pi$  N

03. Ans: (a)

or,



The depth of centre of pressure

$$h_{CP} = \overline{h} + \frac{I_{xx, c}}{A.\overline{h}}$$
  
But  $\frac{I_{xx, c}}{A} = \text{radius of gyration of plate}$   
about centroid  $= \frac{0.22d^2}{4\pi}$  (given)  
 $\overline{h} = \frac{2d}{3\pi}$   
Then,  $h_{CP} = \frac{2d}{3\pi} + \frac{0.22d^2 \times 3\pi}{4\pi \times 2d}$   
 $= \frac{2d}{3\pi} + \frac{0.66d}{8}$   
 $= d\left[\frac{2}{3\pi} + 0.0825\right]$   
 $= 0.295 \text{ d}$   
 $\approx 0.3 \text{d}$  (approximately)

## 04. Ans: (b)

Sol: The capillary rise in a glass tube is given by,

h = 
$$\frac{4\sigma_s}{\gamma \times d}$$
  
=  $\frac{4 \times 0.073}{10^4 \times 0.4 \times 10^{-3}}$  = 0.073 m = 7.3 cm

## 05. Ans: (a)

**Sol:** For an ideal gas,  $P = \rho RT$ 

$$\left(\frac{\partial P}{\partial \rho}\right)_{T} = RT = \frac{P}{\rho}$$
  
or,  $\rho \left(\frac{\partial P}{\partial \rho}\right)_{T} = P$ 

or, coefficient of compressibility for an ideal gas,  $\beta = P$ .

Also, 
$$\beta = \frac{\Delta P}{\Delta \rho / \rho}$$
 Or,  $P = \frac{\Delta P}{\Delta \rho / \rho}$   
Thus,  $\frac{\Delta \rho}{\rho} = \frac{\Delta P}{P}$   
At 10 atm:  $\frac{\Delta \rho}{\rho} = \frac{(11-10)}{10} = 10\%$   
At 1000 atm :  $\frac{\Delta \rho}{\rho} = \frac{1001-1000}{1000} = 0.1\%$   
Ans: (d)

06. Ans: (d)  
Sol: 
$$F_{\rm H} = \gamma_{\rm w} \times \frac{3}{2} \times 3 \times 1 = 4.5 \gamma_{\rm w}$$
  
 $F_{\rm V} = \gamma_{\rm w} \times \frac{\pi}{4} \times 3^2 = 2.25 \pi \gamma_{\rm w}$   
Ratio  $\frac{F_{\rm H}}{F_{\rm V}} = \frac{4.5}{2.25 \pi} = \frac{2}{\pi}$ 



Sol:



Total weight of the arrangement

$$= (1200 + 800) \text{ kN} = 2000 \text{ kN}$$

Volume of sea water displaced,

$$\forall = \frac{2000}{10} = 200 \text{ m}^3$$

and, depth of immersion,  $d = \frac{200}{8 \times 10} = 2.5 \text{m}$ 

Distance of centre of buoyancy (B) from the base point O,

$$OB = \frac{d}{2} = 1.25m$$

Let M be the metacentre, then

BM = 
$$\frac{I}{\forall} = \frac{1}{12} \times \frac{10 \times 8^3}{200} = \frac{32}{15} = 2.13 \text{ m}$$

The metacentric height,

$$GM = BM - BG = BM - (OG - OB)$$
$$= 2.13 - (2.5 - 1.25)$$
$$= 2.13 - 1.25$$
$$= 0.88 \text{ m} \approx 0.9 \text{ m}$$

**08.** Ans: (a)



ESE - 2018 (Prelims) Offline Test Series

:3:

Let the dimension of the block be 'a' and 'b' and, its width be *l*.

Weight of the block = Buoyant force

Or  $a \times b \times l \times 750 \times 9.81 = a \times h \times l \times 0.9 \times 10^3 \times 9.81$ , where h is depth of immersion.

$$h = \frac{750b}{900}$$

Volume of block which is submerged,

$$\mathbf{V}_{\mathrm{s}} = \mathbf{a} \times \mathbf{h} \times \mathbf{l}$$

Total volume of the block =  $a \times b \times l$ 

Volume of block which is not submerged

$$= a \times b \times l - a \times h \times l$$
$$= a \times b \times l - \frac{750 \text{ ab}\ell}{900} = \frac{150}{900} \text{ ab}\ell$$

% of volume of block which is not submerged, is

$$=\frac{ab\ell \times 150/900}{ab\ell} \times 100 = \frac{150}{9} = 16.67\%$$

## 09. Ans: (b)

**Sol:** Weight of filled up balloon + weight of equipment + weight of maximum allowable load = Buoyant force

$$\frac{4}{3}\pi R^{3} \times 0.6 \times g + 50 \times g + Load \times g = \frac{4}{3}\pi R^{3} \times 1.2 \times g$$
  
Or,  $Load = \frac{4}{3}\pi R^{3}(1.2 - 06) - 50$   
 $= \frac{4}{3}\pi \times 3^{3} \times 0.6 - 50$   
 $= 67.86 - 50 = 17.86 \text{ kg}$ 



#### 10. Ans: (d)

### Sol:

- Buoyant force, F<sub>B</sub> = γ∀
   For completely immersed objected F<sub>B</sub> ∝ γ
   So, statement 1 is FALSE.
- $F_R = \gamma \overline{h} A$

First case:

$$\overline{h}_1 = \frac{L}{2}$$
 and  $A = L \times L$   
 $F_{R_1} = \gamma \left(\frac{L}{2}\right) (L \times L) = \frac{\gamma L^3}{2}$ 

Second case:

$$\overline{h}_2 = L + \frac{L}{2} = \frac{3L}{2}$$
 and  $A = L \times L$   
 $F_{R_2} = \gamma \left(\frac{3L}{2}\right) (L \times L) = \frac{3\gamma L^3}{2}$ 

Thus,  $F_{R2} = 3 F_{R1}$ 

So, statement 2 is CORRECT.

•  $\frac{\partial p}{\partial r} = -\rho \omega^2 r$ 

So, statement 3 is FALSE

•  $\frac{dp}{dz} = -\rho g \rightarrow \text{ valid for both incompressible}$ 

and so statement 4 is CORRECT.

## 11. Ans: (a)

**Sol:** The soap bubble will have the greater pressure because there are two surfaces (two surface tension forces) creating the pressure within the bubble. The correct choice is (a).

# 12. Ans: (b)

Sol: From the given stream function

$$\psi = -2x^{2} + y$$

$$u = \frac{-\partial \psi}{\partial y} = -1 \text{ and } v = \frac{\partial \psi}{\partial x} = -4x$$
So, 
$$V = \sqrt{u^{2} + v^{2}} = \sqrt{(1)^{2} + (-4x)^{2}}$$

$$V|_{\substack{x=1\\y=2}} = \sqrt{1 + 16} = \sqrt{17} \text{ m/s} = 4.12 \text{ m/s}$$

Statement (1) is correct.

$$\omega_{z} = \left(\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}\right) = \left(-4, -0\right) = -4 \neq 0$$

Flow is rotational.

Statement (2), is wrong.

Since,  $u = -1 \neq 0$ . So, there is no stagnation point

 $\Rightarrow$  Statement (3) is correct.

## 13. Ans: (c)

Sol: We know that



The maximum vertical rise of the free surface occurs at the back of the tank.

$$\Delta Z_{\text{max}} = \frac{D}{2} \tan \theta = \frac{0.4}{2} \times 0.4 = 0.08 \,\text{m} = 8 \,\text{cm}$$

Therefore, the maximum initial water height in the tank to avoid spilling is

$$h_{max} = 60 - 8 = 52 \text{ cm}$$

## 14. Ans: (b)

Sol: Normal acceleration,

$$a_{\rm N} = \frac{V^2}{R} = g$$
 (given)  
So,  $V = \sqrt{gR} = \sqrt{9.81 \times 0.6} = 2.42$  m/s

## 15. Ans: (b)

Sol: It is known for the trajectory of the jet that  $x = V_o \times t$  where  $V_o$  is the velocity of jet at the orifice

And 
$$y = \frac{1}{2}gt^2 = \frac{1}{2}g\left(\frac{x}{V_o}\right)^2$$

From the given coordinates, i.e., x = 1m and y = 1m,

$$V_o = 1 \left(\frac{10}{2 \times 1}\right)^{1/2} = 5^{1/2} = 2.21 \text{ m/s}$$

## 16. Ans: (a)

Sol: In the x-y plane, the rate of rotation of a fluid particle is given by

$$\omega_{z} = \frac{1}{2} \left( \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right) = \frac{1}{2} \left( 0 - \frac{v_{0}}{h} \right)$$
$$= -\frac{v_{0}}{2h}$$

The other components will be zero.





Sol: The equation of the streamlines for the given

flow field is given by

$$\frac{dy}{dx} = -\frac{2Cxy}{C(x^2 - y^2)} = \frac{-2xy}{x^2 - y^2} = \frac{v}{u}$$

For flow to be parallel to y-axis, u = 0

Or, 
$$\frac{dy}{dx} = \frac{-2xy}{x^2 - y^2} = \infty$$

This is possible when  $x = \pm y$ 

#### 18. Ans: (b)

**Sol:** The two dimensional continuity equation in cylindrical polar coordinates is

$$\frac{1}{r}\frac{\partial}{\partial r}(rv_{r}) + \frac{1}{r}\frac{\partial v_{\theta}}{\partial \theta} = 0$$
  
Or,  $\frac{\partial v_{\theta}}{\partial \theta} = -\frac{\partial}{\partial r}(rv_{r})$   
For  $v_{r} = \frac{r}{2} + \frac{r^{2}}{3}\sin\theta$   
 $\frac{\partial v_{\theta}}{\partial \theta} = \frac{-\partial}{\partial r}\left(\frac{r^{2}}{2} + \frac{r^{3}}{3}\sin\theta\right) = -r - r^{2}\sin\theta$ 

Integrating the above equation with respect to  $\theta$ ,

we get,  $v_{\theta} = -r(\theta - r\cos\theta) + f(r)$ 

#### 19. Ans: (c)

Sol: The velocity of water in the pipe is given by,

$$V = \sqrt{\frac{2(P_0 - P)}{\rho}} = \sqrt{2g(h_0 - h)}$$

where  $P = \rho gh$ =  $\sqrt{2 \times 10(0.35 - 0.2375)}$ =  $\sqrt{20 \times 0.1125} = \sqrt{2.25} = 1.5 m/s$ 

#### 20. Ans: (a)

Sol:  $P + \gamma Z + \frac{\rho V^2}{2} \rightarrow$ This is the Bernoulli equation per unit volume of the fluid flowing.

$$\frac{P}{Q} + gZ + \frac{V^2}{2} \rightarrow$$
 This is Bernoulli equation

per unit mass of the fluid flowing.

$$\frac{P}{\gamma} + Z + \frac{V^2}{2g} \rightarrow$$
 This is the Bernoulli

equation

## 21. Ans: (a)

Sol: Efficiency of transmission

 $= \frac{\text{Power delivered at the pipe outlet}}{\text{Power available at the pipe inlet}}$ 

$$=\frac{50\times10^{3}}{10^{4}\times12\times10^{-3}\times500}\times100$$
$$=\frac{1000}{12}=83.3\%$$

22. Ans: (d)

Sol: We know that  $\Delta P = \rho L \frac{\Delta V}{\Delta t}$   $250 \times 10^3 = 10^3 \times 2500 \times \frac{2}{\Delta t}$  $\Delta t = \frac{10^3 \times 2500 \times 2}{250 \times 10^3} = 20 s$ 



**Sol:** The jet is deflected by 180°. So, the force exerted by the jet on the vane is

$$F = 2\rho A_{jet} V^{2} + \rho A_{vane} L \times g$$
  
= 2 × 10<sup>3</sup> ×  $\frac{\pi}{4}$  × 4 × 10<sup>-4</sup> × 10<sup>2</sup> + 10<sup>3</sup> ×  $\frac{\pi}{4}$  × (2 × 10<sup>-2</sup>)<sup>2</sup> × 1 × 10  
= 21  $\pi$  N

#### 24. Ans: (c)

Sol: For incompressible flow,  $\rho$  = constant. So,

 $\frac{\partial \rho}{\partial t} = 0$  even for unsteady flow. So, statement (i) is correct. The second statement is also correct.

#### 25. Ans: (b)

**Sol:** If there is no hole in the vane then the force acting on the vane is

$$F = \rho A V^2 (1 + \cos \theta)$$

Due to hole effective area of the deflected jet reduces to (A–a)

$$\therefore F = \rho(A-a)V^{2}(1+\cos\theta)$$
$$= \rho(A-a)V^{2}\left(1+\frac{1}{2}\right) \qquad [as \ \theta = 60^{\circ}]$$
$$= \frac{3}{2}\rho(A-a)V^{2}$$

#### 26. Ans: (c)

**Sol:** Considering force F (towards right) to be force exerted on the fluid by contraction, the linear moment equation may be written as:

$$F_{\text{on fluid}} - P_1 A_1 + P_2 A_2 = \rho A_1 V_1 (-V_2 + V_1)$$

$$F_{\text{on fluid}} = (P_1 A_1 - P_2 A_2) - \rho A_1 V_1^2 \left(\frac{V_2}{V_1} - 1\right)$$

$$F_{\text{on fluid}} = \left(P_1 A_1 - P_2 A_2\right) - \rho A_1 V_1^2 \left(\frac{A_1}{A_2} - 1\right)$$

Thus,  $F_{on contraction} = F_{on fluid}$  but acting towards left.

#### 27. Ans: (c)

**Sol:** Re = 1600

So, the flow is laminar and

$$f = \frac{64}{Re} = \frac{64}{1600} = 0.04$$

Head loss, 
$$h_f = \frac{fLV^2}{2gD}$$

Hence, 
$$V = \sqrt{\frac{h_f \times 2gD}{fL}}$$

$$= \sqrt{\frac{2 \times 2 \times 10 \times 0.25 \times 10^{-2}}{0.04 \times 5}}$$
$$= \sqrt{0.5} = 0.707 \text{ m/s}$$

#### 28. Ans: (a)

Sol: The energy equation gives :

$$\Delta z = \frac{f L V^2}{2gd}$$
$$d = \frac{f L V^2}{2g \Delta z}$$
$$= \frac{0.04 \times 180 \times 2.5^2}{2 \times 10 \times 90}$$
$$= 0.025 \text{ m} = 2.5 \text{ cm}$$



#### 29. Ans: (b)

Sol: Blower power =  $\rho g Q h_f$ 1200 = 1.2 × 10 × 800Q<sup>3</sup>

$$Q^{3} = \frac{1200}{1.2 \times 10 \times 800}$$
$$\Rightarrow Q = \frac{1}{2} \text{ m}^{3}/\text{s}$$

30. Ans: (c)

Sol: 
$$C_1 = \sqrt{\frac{K_1}{\rho_1}}$$
,  $C_2 = \sqrt{\frac{K_2}{\rho_2}}$   
 $\frac{C_1}{C_2} = \sqrt{\frac{K_1}{K_2}} \sqrt{\frac{\rho_2}{\rho_1}}$   
 $= \sqrt{\frac{2.16 \times 10^9}{1.296 \times 10^9}} \times \sqrt{0.6} = 1$ 

#### 31. Ans: (d)

- **Sol:** Losses for flow through valves and fittings are expressed in terms of loss coefficient K<sub>s</sub>. It is expressed in terms of equivalent length of a straight pipe
- 32. Ans: (c)

Sol: 
$$C = \sqrt{\frac{K}{\rho}}$$
  
 $\Delta P = \rho C V$   
 $\rho g H = \rho \times \sqrt{\frac{K}{\rho}} \times V$   
 $H = \frac{V}{g} \times \sqrt{\frac{K}{\rho}}$ 

33. Ans: (b) Sol:  $h = \frac{P_1}{\rho g} - \frac{P_2}{\rho g} = \frac{1}{10^3 \times 10} (30 - 10) \times 10^4$  = 20 m  $Q = C_d \times \frac{a_1 a_2}{\sqrt{a_2^2 - a_2^2}} \times \sqrt{2gh}$   $= 0.6 \times \frac{250 \times 150 \times 10^{-4}}{\sqrt{(250)^2 - (150)^2}} \times \sqrt{2 \times 10 \times 20}$   $= 0.6 \times \frac{250 \times 150 \times 10^{-4}}{200} \times 20$ Q = 225 lit/sec

Sol: 
$$V = \frac{1}{A} \int_{0}^{r_{0}} u_{max} \left(1 - \frac{r}{r_{0}}\right)^{m} 2\pi r dr$$
$$= 2\pi \times \frac{u_{max}}{\pi r_{0}^{2}} \int_{0}^{r_{0}} \left(1 - \frac{r}{r_{0}}\right)^{m} r dr$$
$$= \frac{2u_{max}}{r_{0}} \int_{0}^{r_{0}} \left(1 - \frac{r}{r_{0}}\right)^{m} \left(\frac{r}{r_{0}}\right) dr$$
$$\left(taking\left(1 - \frac{r}{r_{0}}\right) = z\right)$$
$$= \frac{2u_{max}}{r_{0}} \int_{0}^{1} (z)^{m} (1 - z) r_{0} dz$$
$$= \frac{2u_{max}}{r_{0}} \times \frac{r_{0}}{(m+1)(m+2)}$$
$$= 2u_{max} \frac{1}{(m+1)(m+2)}$$



#### 35. Ans: (b)

**Sol:** In Pipe flow, shear stress is zero at the centre and increases linearly from centre to the pipe wall.

$$\Delta P = \frac{128\mu QL}{\pi d^4}$$
$$Q \propto \frac{1}{\mu}$$

In a constant diameter pipe pressure decreases linearly.  $\left(\frac{P}{\rho g} + Z\right)$  also decreases

linearly.

#### 36. Ans: (b)

**Sol:**  $\mu_{water}$  at 40° C is 0.6531 mPa-s

## 37. Ans: (a)

Sol: 
$$h_{L_1} = \frac{32\mu V(2L)}{\rho g D^2}$$
  
 $h_{L_2} = 12 \times \frac{\mu V L}{\rho g B^2}$   
 $\frac{h_{L_1}}{h_{L_2}} = \frac{32}{12} \times \frac{2}{1} \times \frac{B^2}{D^2}$   
 $\frac{32}{12} \times 2 \times \frac{3^2}{12^2} = \frac{1}{3}$ 

#### 38. Ans: (b)

Sol: Let  $z_1$  and  $z_2$  be the elevation of reservoir and nozzle

Let 'D' diameter of pipe and 'd' be diameter of nozzle.

Net head = 
$$(z_1-z_2) = H$$
  
 $z_1 + 0 + 0 = z_2 + \frac{V_2^2}{2g} + h_L$   
 $h_L = 0.5 \frac{V^2}{2g} + f \cdot \frac{L}{D} \cdot \frac{V^2}{2g}$ 

(entrance loss + pipe loss)

$$V_{2} = \left(\frac{D}{d}\right)^{2} V$$

$$z_{1} - z_{2} = \left(\frac{D}{d}\right)^{4} \cdot \frac{V^{2}}{2g} + 0.5 \frac{V^{2}}{2g} + f \cdot \frac{L}{D} \cdot \frac{V^{2}}{2g}$$

$$f = 0.07, D = 2 \text{ cm}, d = 1 \text{ cm}$$

$$H = \left[0.5 + f \cdot \frac{L}{D} + \left(\frac{D}{d}\right)^{4}\right] \cdot \frac{V^{2}}{2g}$$

$$= \left[0.5 + 7 \times 10^{-2} \frac{1}{2 \times 10^{-2}} + 16\right] \cdot \frac{V^{2}}{2g}$$

$$= [0.5 + 3.5 + 16] \times \frac{16}{20}$$

$$H = 16$$

### **39.** Ans: (a)

Sol: The specific speed values given actually refer to the non-dimensional specific speed (in rad/s<sup>-1</sup>). Here, again the minimum value (0.1) is for Pelton wheel, followed by Francis turbine (1.0). The propeller turbine will have the value of 3.0. Kaplan turbine will have the highest value of 4.0.



**Sol:** 
$$H_m = \frac{P_d - P_s}{\rho g} + \frac{V_d^2 - V_s^2}{2g} + Z_d - Z_s$$

Assuming suction and delivery pipes of same diameter and neglecting elevation changes between inlet and exit.

$$H_{m} = \frac{P_{d} - P_{s}}{\rho g}$$
$$= \frac{300 \times 10^{3} - (-25 \times 10^{3})}{1000 \times 10} = 32.5 \text{ m}$$

41. Ans: (c)

Sol:



For parallel combination, head remains same but discharge becomes twice.

i.e., 
$$H_P = H \& Q_P = 2Q$$
  
let  $H_p = a - b Q_p^2$   
i.e.,  $H = a - b (2Q)^2$   
 $40 - 60 Q^2 = a - 4b Q^2$   
Comparing both sides  
 $a = 40 \& b = 15$   
 $\therefore H_p = 40 - 15 Q^2$ 

Sol: 
$$\frac{D_m}{D_p} = \frac{1}{5}$$
,  $\frac{H_m}{H_p} = \frac{4}{1}$   
 $P \propto QH$   
 $Q \propto D^2 \sqrt{H}$   
 $\therefore P \propto D^2 H^{3/2}$   
 $\frac{P_p}{P_m} = \left(\frac{D_p}{D_m}\right)^2 \times \left(\frac{H_p}{H_m}\right)^{3/2}$   
 $= \left(\frac{5}{1}\right)^2 \times \left(\frac{1}{4}\right)^{3/2} = \frac{25}{8} = 3.125$ 



Admissions are open at all our centers

H. O. : Hyderabad : Ph : 040-23234418,19,20

Bangalore		Kukatpally		Delhi		Bhopal		Pune		Bhubaneswar	
9341299966		040-6597 4465		9205282121		0755-2554512		020-25535950		0674-2540340	
Lucknow	Patna		Chennai		Vijayawada		Vishakapatnam		Tirupathi		Kolkata
808199966	9308699966		044-42123289		0866-2490001		0891-6616001		0877-2244388		8297899966



#### 43. Ans: (d)

Sol: Velocity head in delivery pipe is maximum at the middle of delivery stroke

i.e.,  $\theta = 270^{\circ}$ 

#### 44. Ans: (d)

**Sol:** The velocity variation in suction pipe of a single acting reciprocating pump is



45. Ans: (d)

Sol: 
$$Q = \frac{ALN}{60}$$
,  $L = 2r$   
 $\therefore Q \propto r$   
 $\propto N$   
 $\frac{Q_2}{Q_1} = \left(\frac{r_2}{r_1}\right) \times \left(\frac{N_2}{N_1}\right) = \left(\frac{1}{2}\right) \times \left(\frac{2}{1}\right) = 1$ 

$$\frac{Q_2 - Q_1}{Q_1} \times 100 = 0\%$$

### 46. Ans: (c)

Sol: Air vessel on delivery side does not have any effect on suction side. It reduces velocity fluctuations in delivery side. Hence, acceleration head is reduced only on delivery side.

> Static delivery head is the elevation difference between delivery pipe exit and pump exit. It is independent of any other parameter.

## 47. Ans: (b)

Sol: 
$$\eta_{max} = \frac{1 + \cos \theta}{2} = \frac{1 + \cos(180 - 120)}{2} = 0.75$$
  
Applying energy balance to the rotor  
 $\frac{V_1^2}{2g} = H_e + \frac{V_2^2}{2g}$   
 $= \eta_{max} \times \frac{V_1^2}{2g} + \frac{V_2^2}{2g}$   
 $\frac{V_2^2}{2g} = (1 - \eta_{max}) \frac{V_1^2}{2g}$   
 $\frac{V_2^2/2g}{V_1^2/2g} = (1 - \eta_{max}) = 0.25$ 

48. Ans: (b) Sol: F = ρaV (V - u) (1 + cos θ)At the starting u = 0

ACE Engineering Academy



$$\therefore F = \rho a V^{2} (1 + \cos \theta)$$
$$T = F \times R = \rho a V^{2} (1 + \cos \theta) \times \frac{D}{2}$$
$$= 1000 \times (0.1) \times 50 \times (1 + \cos 60) \times \frac{2}{2}$$
$$= 7.5 \text{ kN.m}$$

#### 49. Ans: (a)

**Sol:**  $T = F \times R$ 

$$\frac{T_2}{T_1} = \frac{F_2}{F_1}$$
  
i.e.,  $\frac{1}{2} = \frac{\rho a V_2 (V_2 - u_2) (1 + k \cos \theta)}{\rho a V_1 (V_1 - u_1) (1 + k \cos \theta)}$ 

The jet speed only depends on net head. hence  $V_1 = V_2 = V$ 

For maximum efficiency  $\frac{u}{V} = \frac{1}{2}$ 

The initial condition corresponds to maximum efficiency

$$u_{1} = \frac{V}{2}$$

$$\frac{1}{2} = \frac{V - u_{2}}{\left(V - \frac{V}{2}\right)}$$
i.e.,  $\frac{V}{4} = V - u_{2}$ 

$$u_{2} = \frac{3V}{4}$$

$$\frac{u_{2} - u_{1}}{u_{1}} = \frac{\frac{3V}{4} - \frac{V}{2}}{\frac{V}{2}} \times 100 = 50\%$$

#### 50. Ans: (d)

**Sol:** Whether, the Francis turbine has radial vanes or not, the velocity of flow is always radial. If the rotor blade at inlet is radial then the relative velocity (which is tangential to the blade) is also radial.

#### 51. Ans: (a)

Sol:  

$$\eta_{h} = \frac{u_{1}V_{w1}}{gH}$$
i.e.,  $0.8 = \frac{10 \times V_{w1}}{10 \times 12}$  [as  $g = 10 \text{ m/s}^{2}$ ]  
 $V_{w1} = 9.6 \text{ m/s}$   
 $\tan \alpha_{1} = \frac{V_{f1}}{V_{w1}} = \frac{5}{9.6} = 0.52$   
 $\tan 30^{\circ} = \frac{1}{\sqrt{3}} = \frac{\sqrt{3}}{3} = \frac{1.732}{3} = 0.577$   
 $0.52 < 0.577 \Rightarrow \alpha_{1} < 30$   
 $\therefore \alpha_{1} = 28^{\circ}$   
The exact value of  $\alpha_{1}$  is

ne exact value of  $\alpha_1$  is

$$\alpha_1 = \tan^{-1} \left( \frac{5}{9.6} \right) = 27.5^{\circ}$$

## 52. Ans: (b)

Sol: 
$$Q = A_f V_f = \frac{\pi}{4} (D_t^2 - D_h^2) \times V_f$$
  
=  $\frac{\pi}{4} (3^2 - (\frac{1}{3} \times 3)^2) \times 5$   
=  $\frac{\pi}{4} \times (9 - 1) \times 5$   
=  $10 \pi m^3/s = 31.4 m^3/s$ 

ACE Engineering Academy



#### 53. Ans: (a)

**Sol:**  $\tan \alpha_1 = \frac{V_{f1}}{V_{w1}}$ 

In Kaplan turbine  $V_{\rm w1}$  follows free vortex distribution

$$V_{w1} = \frac{k}{r}$$

Thus  $V_{w1}$  decreases from hub towards tip. As  $V_f$  is constant  $\alpha_1$  increases from hub towards tip.

54. Ans: (c)

Sol: 
$$N_s = \frac{N\sqrt{P}}{H^{5/4}}$$
  
 $P = F \times V = F \times \frac{L}{T} = F L T^{-1}$   
 $N = T^{-1}$   
 $N_s = \frac{T^{-1}\sqrt{F L T^{-1}}}{L^{5/4}}$   
 $= F^{1/2} I^{-3/4} T^{-3/2}$ 

55. Ans: (a)

Sol: 
$$F_{D} = \frac{C_{D}}{2} \rho U_{\infty}^{2}$$
$$\therefore \frac{F_{D2}}{F_{D1}} = \frac{C_{D2}}{C_{D1}} \times \frac{\rho_{2}}{\rho_{1}} \{\because U_{\infty_{1}} = U_{\infty_{2}}\}$$
$$= \sqrt{\frac{Re_{1}}{Re_{2}}} \times \frac{\rho_{2}}{\rho_{1}} \{\because C_{D} = \frac{1.328}{\sqrt{Re}}\}$$
$$= \sqrt{\frac{\rho_{1}}{\rho_{2}}} \times \sqrt{\frac{\mu_{2}}{\mu_{1}}} \times \frac{\rho_{2}}{\rho_{1}}$$

$$= \sqrt{\frac{\mu_2}{\mu_1}} \times \sqrt{\frac{\rho_2}{\rho_1}}$$
  
i.e.  $F_{Dr} = \sqrt{\mu_r} \times \sqrt{\rho_r}$ 

#### 56. Ans: (c)

Sol: The displacement thickness  $(\delta^*)$  can be considered as an imaginary growth in the thickness of the wall so that the outside flow can be considered as uniform flow.



Using continuity equation

$$U_1 \pi R^2 = U_2 \pi \left( R - \delta^* \right)^2$$
$$\frac{U_2}{U_1} = \left( \frac{R}{R - \delta^*} \right)^2$$

57. Ans: (b)

Sol: 
$$u = U_{\infty} \left[ \frac{3}{2} \frac{y}{\delta} - \frac{1}{2} \left( \frac{y}{\delta} \right)^3 \right]$$
  
$$\frac{du}{dy} = U_{\infty} \left[ \frac{3}{2} \frac{1}{\delta} - \frac{3}{2} \left( \frac{y}{\delta} \right)^2 \times \frac{1}{\delta} \right]$$
$$\frac{du}{dy} \Big|_{y=\frac{\delta}{2}} = \frac{U_{\infty}}{\delta} \left[ \frac{3}{2} - \frac{3}{2} \left( \frac{1}{2} \right)^2 \right] = \frac{9}{8} \frac{U_{\infty}}{\delta}$$
$$\tau = \mu \frac{du}{dy} = \frac{9}{8} \frac{\mu U_{\infty}}{\delta}$$
$$\therefore \quad k = \frac{9}{8}$$

ACE Engineering Academy



### 58. Ans: (d)

**Sol:** Velocity gradient normal to the surface is very high as compared to streamwise velocity gradient.

i.e., 
$$\frac{\partial u}{\partial y} >> \frac{\partial u}{\partial x}$$

Continuity equation must be satisfied at each point in a flow

i.e., 
$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$
  
or  $\frac{\partial u}{\partial x} = -\frac{\partial v}{\partial y}$   
 $\therefore \left| \frac{\partial u}{\partial x} \right| = \left| \frac{\partial v}{\partial y} \right|$ 

### 59. Ans: (d)

**Sol:** All of the above methods will reduce the chances of separation. Rough surface makes the boundary layer turbulent and the turbulent boundary layer is comparatively less susceptible to separation.

## 60. Ans: (c)

Sol: 
$$\delta(x) = \frac{5x}{\sqrt{Re_x}} = \frac{5x}{\sqrt{\frac{\rho U_{\infty} x}{\mu}}} \propto \sqrt{x}$$
$$C_{fx} = \frac{0.664}{\sqrt{Re_x}} = \frac{0.664}{\sqrt{\frac{\rho U_{\infty} x}{\mu}}} \propto \frac{1}{\sqrt{x}}$$
$$\therefore \frac{\delta(x)}{C_{fx}} \propto \frac{\sqrt{x}}{\left(\frac{1}{\sqrt{x}}\right)} \propto x$$





#### ★ HIGHLIGHTS ★

- Detailed solutions are available.
- All India rank will be given for each test.
- Comparison with all India toppers of ACE students.



### ESE - 2018 (Prelims) Offline Test Series

#### 61. Ans: (c)

**Sol:** Normally a specific property of a substance is defined per unit mass. But for a fluid its specific weight is defined as weight per unit volume. It is an intensive property. So, statement (II) is wrong.

#### 62. Ans: (d)

**Sol:** The coefficient of volume expansion of a fluid differs from its bulk modulus of compressibility. So, statement (I) is wrong. However, statement (II) is correct.

#### 63. Ans: (b)

**Sol:** Reynolds number is the ratio of inertia forces to the viscous forces in the fluid flow. These inertia forces generate due to the change in momentum of the fluid. At higher Reynolds number, flow is turbulent and the velocity profile is logarithmic.

## 64. Ans: (a)

**Sol:** For irrotational flow, Bernoulli equation can be applied at any two points in the flow. Centrifugal forces will cause the increase in pressure from inside to outside edge. Application of Bernoulli equation will show the decrease of velocity from inside to outside edge.

#### 65. Ans: (a)

**Sol:** For stability of floating bodies, GM > 0. So, if a body with wide rectangular cross section is tilted slightly, its B (centre of

buoyancy) shifts more towards the tilted end resulting in larger BM. So, GM will be positive. Also it provides a righting couple. Therefore, the body will be in the stable condition.

## 66. Ans: (a)

:15:

Sol:  $C_d$  of venture meter lies between 0.97 to 0.99. This is mainly due to smooth gradual variation of cross-sectional area down to throat and then back to the full pipe area. This results in less friction losses. However, in the case of orifice meter there is an abrupt constriction at the orifice plate. This causes the formation of eddies which leads to higher friction losses. The  $C_d$  of orifice meter varies from 0.58 to 0.65.

## 67. Ans: (c)

Sol: The resultant hydrostatic force is given by  $F_R = \gamma \overline{h} A.$ 

So,  $F_R$  can be determined if  $\overline{h}$  and A are known. So, statement (I) is correct. However,  $F_R$  depends on specific weight of the liquid in which it is submerged. So, statement (II) is wrong.

## 68. Ans: (d)

**Sol:** In steady flow, local acceleration is zero but convective acceleration may not be zero.



**Sol:** The unsymmetric pressure distribution on the football surface is due to Magnus effect. Spinning of football generates low velocity on one side and high velocity on other side.



#### 70. Ans: (c)

Sol: At separation point shear stress is zero because velocity gradient becomes zero. Pressure gradient is positive when flow is about to separate.

#### 71. Ans: (c)

**Sol:** For blunt object pressure drag is major contributor to the overall drag. Streamlining avoids the flow separation hence pressure drag is reduced. Therefore, overall drag is also reduced.

Streamlining in fact increases the skin friction drag as the contact area is increased. Hence, Statement (II) is false.

## 72. Ans: (a) Sol:



Hence kinetic energy head loss is reduced by  $\frac{V_2^2 - V_3^2}{2\sigma}$ 

# 73. Ans: (b)

Sol: Kaplan turbine has higher efficiency as compared to propeller turbine at part load condition because Kaplan turbine has adjustable rotor blades where as propeller turbine has fixed rotor blades. Due to adjustable rotor blades, the Kaplan turbine can adjust to different flow condition with minimum separation loss.

## 74. Ans: (d)

**Sol:** Statement (II) is the definition of NPSH. Hence, it is correct statement.

NPSH = 
$$\frac{P_i}{\rho g} + \frac{V_i^2}{2g} - \frac{P_v}{2g}$$
  
=  $H_a - H_s - H_v - h_f$ 

From above equation it is clear that as suction head increases, NPSH decreases. Hence, statement I is wrong.

## 75. Ans: (d)

Sol: Theoretical head developed by the pump is

$$H_e = \frac{u_2 V_{w2}}{g}$$

Head developed only depends on exit velocity triangle. Thus statement (I) is wrong.

The theoretical pressure rise across the pump is  $\rho g H_e = \rho u_2 V_{w2}$ .

Hence, statement (II) is correct.

# **GATE TOPPERS**



**ESE TOPPERS** 





# 204, Rahman Plaza, Abids, Hyderabad Ph : 040-23234418/19/20